

Royal Netherlands Meteorological Institute Ministry of Infrastructure and the Environment

OMI and AURA

Status
Instrument, Spacecraft
and Operations

MOWG College Park, 16 September 2014

Presentation content



- Instrument status
- Spacecraft status
- Operations status



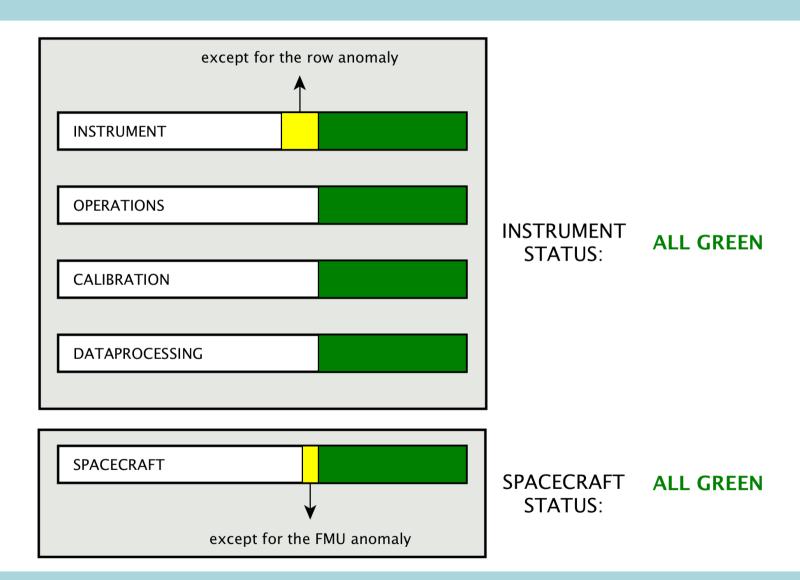
OMI flight hardware

Focus is on those items that can potentially impact the quality of the science data.



Overall current status





Overall status



Anomalies since launch

Only <u>three</u> anomalies since launch that impacted the science data:

- Instrument
 - 1) Folding Mirror Mechanism anomaly in spring 2006: <u>status</u> solved
 - 2) Row anomaly since May 2008: <u>status ongoing</u> but appropriate L1B flagging scheme (identifying affected groundpixels) has been implemented
- Spacecraft
 - 3) Formatter/Multiplexer Unit anomaly since Dec. 2007: status ongoing but work-around has been implemented. No impact on the OMI science data.



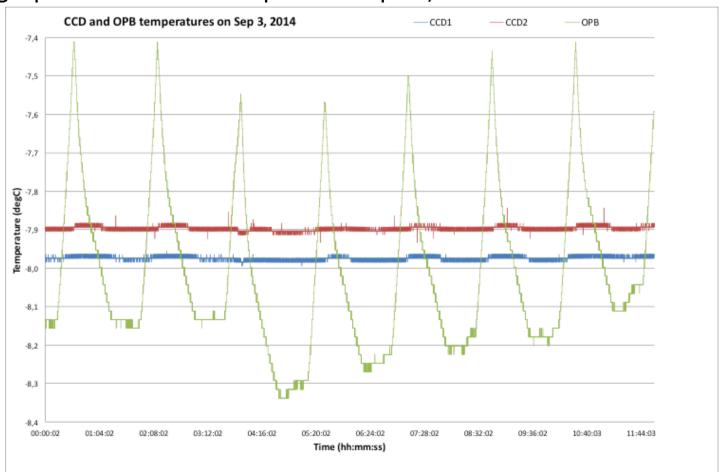
Summary

- Instrument performs nominal (with exception of row anomaly)
- No other instrument anomalies
- CCD temperatures still very stable
- All three mechanisms behave nominally
- Life limited items (mechanisms, internal calibration source) within budget



CCD Temperatures

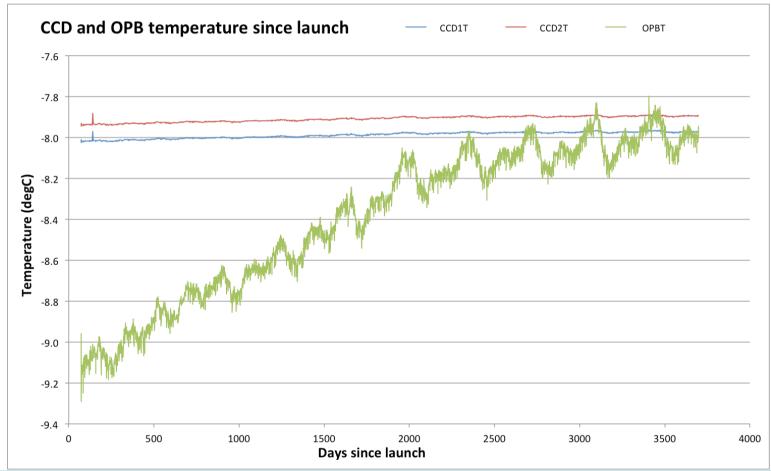
• Thermal controller keeps the CCD temperatures constant in a temperature varying optical bench. Example for Sep. 2, 2014.





CCD Temperatures

• Due to the degradation of the radiator, the OPB temperature has increased over the mission, but the CCD temperatures remained almost constant.





Thermal Stability

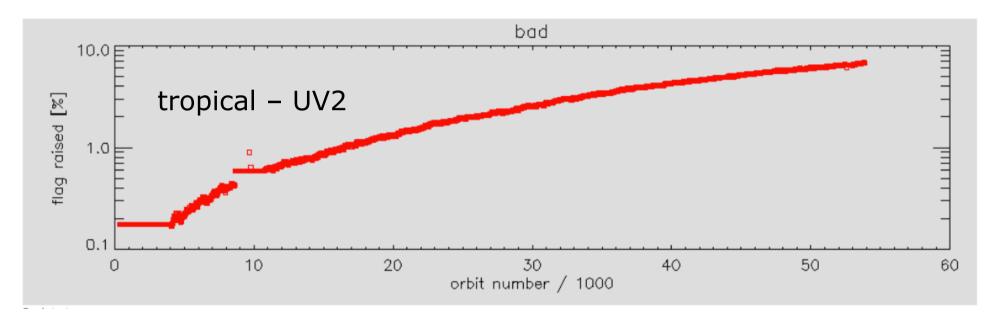
Overall thermal stability very good; thermal settings have not changed

	2004	2014	trend
Optical bench	264.0 K	265.1 K	+1.1 K
UV channel CCD	265.13 K	265.18 K	+0.05 K
VIS channel CCD	265.21 K	265.26 K	+0.05 K
UV electronics ELU	290.0 K	290.7 K	+0.7 K
VIS electronics ELU	291.5 K	292.1 K	+0.6 K
AUX electronics ELU	289.5 K	290.2 K	+0.7 K
UV CCD ATC PWM	20 %	6 %	-14. %
VIS CCD ATC PWM	24 %	10 %	-14 %



CCD Radiation Damage

CCD detector radiation damage (percentage of pixels flagged in L1b)

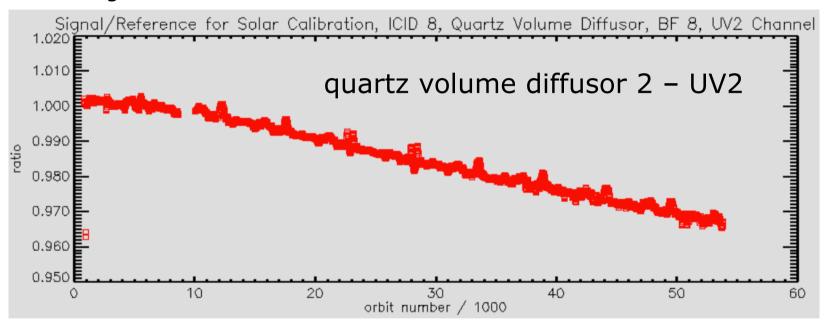


	tropical		midlatitude			arctic			
	UV1	UV2	VIS	UV1	UV2	VIS	UV1	UV2	VIS
bad	8%	7%	9%	18%	7%	9%	25%	7%	9%
RTS	5%	6%	5%	5%	6%	6%	6%	6%	6%



Optical Degradation

 Optical degradation extremely low: no other UVN-like instruments with such low degradation

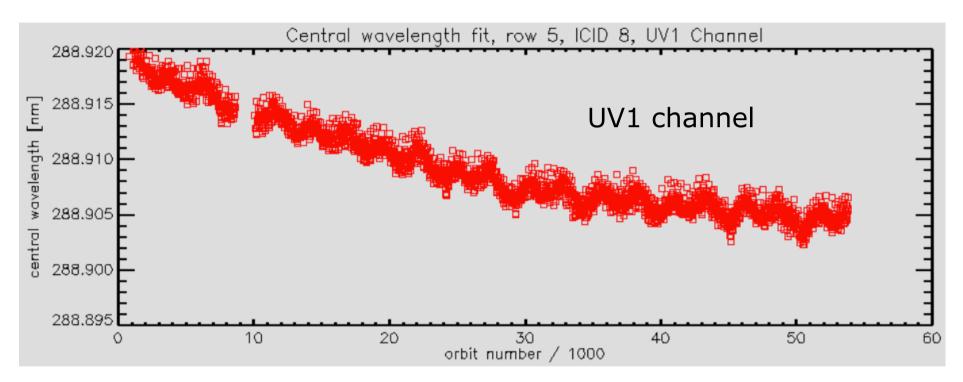


	UV1	UV2	VIS
QVD path (daily)	0.945	0.967	0.974
ALU1 path (weekly)	0.972	0.981	0.980
ALU2 path (monthly)	0.980	0.985	0.984
optical	-2.0 %	-1.5 %	-1.6 %
QVD	-3.5 %	-1.8 %	-1.0 %



Spectral Stability

• Spectral stability irradiance



	UV1	UV2	VIS
trend	-0.015 nm	0.000 nm	+0.002 nm
seasonal [pp]	0.002 nm	0.002 nm	0.002 nm



Life Limited Items

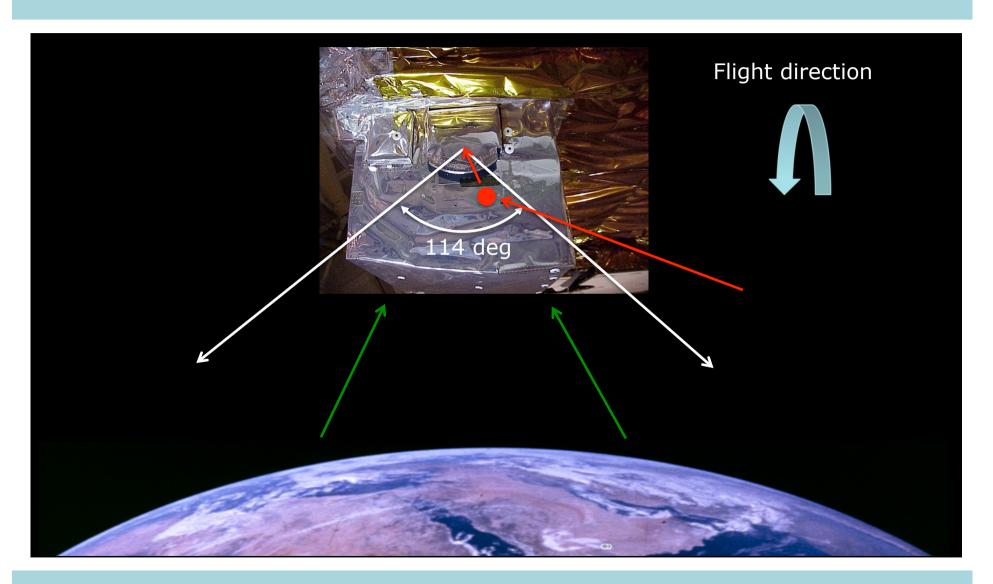
- All three mechanisms and internal calibration source are far within budget
- Remaining budget, assuming unchanged operations baseline:
 - > Solar Aperture Mechanism: 56 years
 - ➤ Diffuser Mechanism: 43 years
 - > Folding Mirror Mechanism: 9 years
 - ➤ White Light Source: 19 years

Description	SAM Cycles	DifM Cycles	FMM Cycles	WLS Seconds
Total Used	6985	7717	18145	108441
Budget	30916	30916	30916	216000
Budget Spent	22.59%	24.96%	58.69%	50.20%

Current budget (status Sep 3, 2014)



Row anomaly: cause





Row anomaly: errors

Errors in L1b caused by the row anomaly

- A multiplicative error
 Caused by the partial blockage of the nadir field of view resulting in reduced radiance levels for specific rows.
- A wavelength shift
 Caused by inhomogeneous illumination of the spectral slit due to the blocking material, resulting in a change of the slit function.
- Stray earthlight related additive error
 Caused by earthlight, reflected by the blocking material from outside the OMI fov into the nadir port.
- Stray sunlight related additive error
 Caused by sunlight, reflected by the blocking material into the nadir port for part of the orbit.



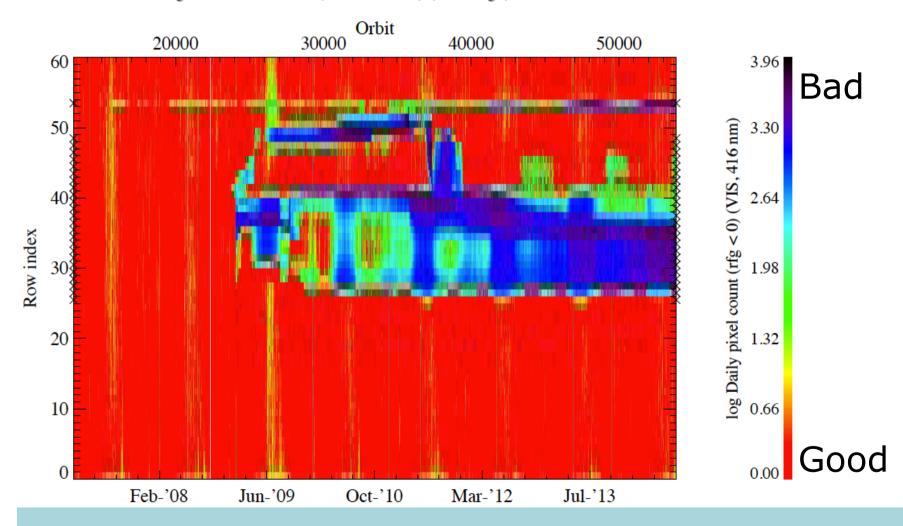
Row anomaly: status

- Affected ground pixels are flagged in the L1b product
- Row anomaly behaviour continues to change on short-term as well as long-term timescales (see next slide).
- The row anomaly is monitored on a daily basis using L1b data. Tools have been developed for this.
- Row Anomaly Monitoring (RAM) reports are generated daily and distributed to Dutch and US core team members.
- In case needed, L1b and L2 data are post-processed: update
 of the flags identifying affected rows. Flag definition needs
 approval from the OMI core team members.



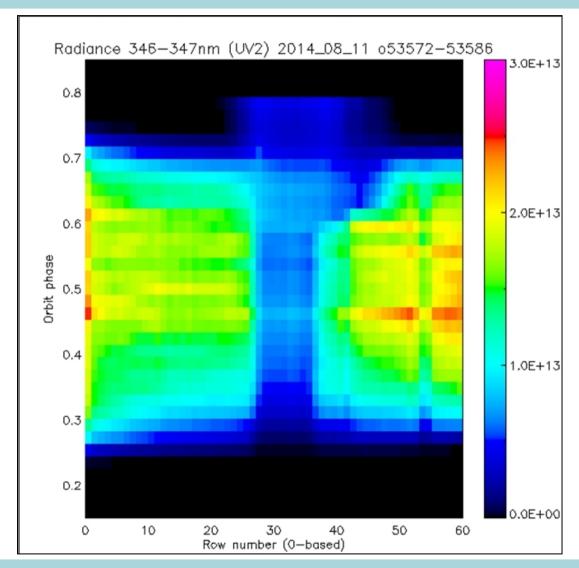
Row anomaly: monitor

Number of negative reflectances (VIS, 416 nm) (Blockage)





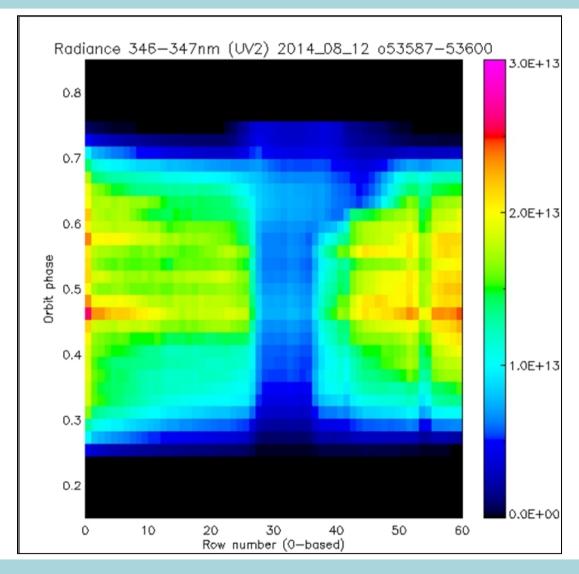
Row anomaly change on August 13, 2014



UV2 radiance daily average



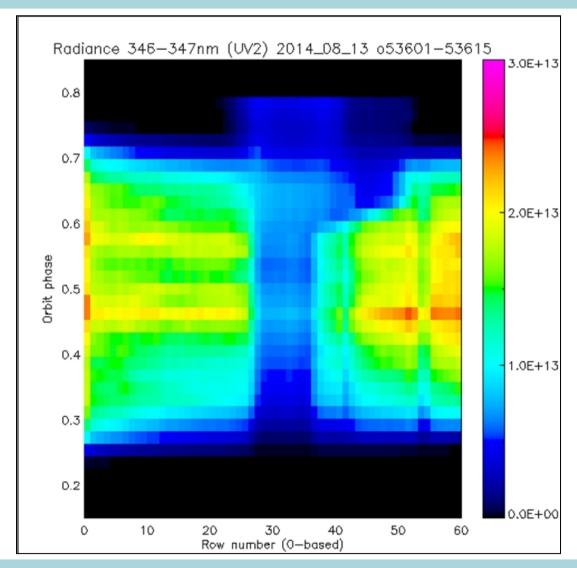
Row anomaly change on August 13, 2014



UV2 radiance daily average



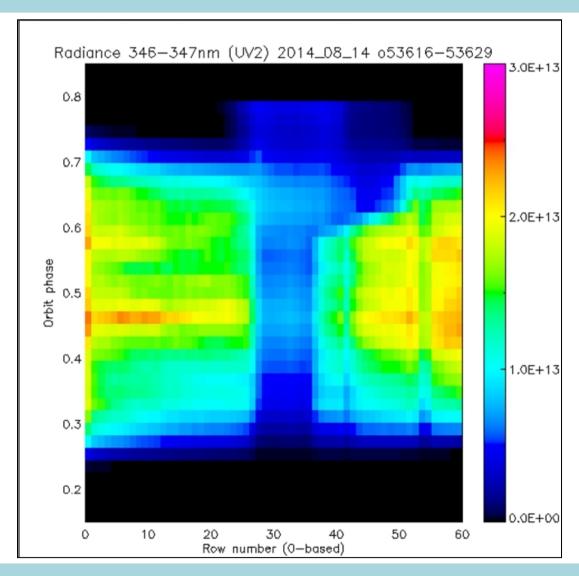
Row anomaly change on August 13, 2014



UV2 radiance daily average



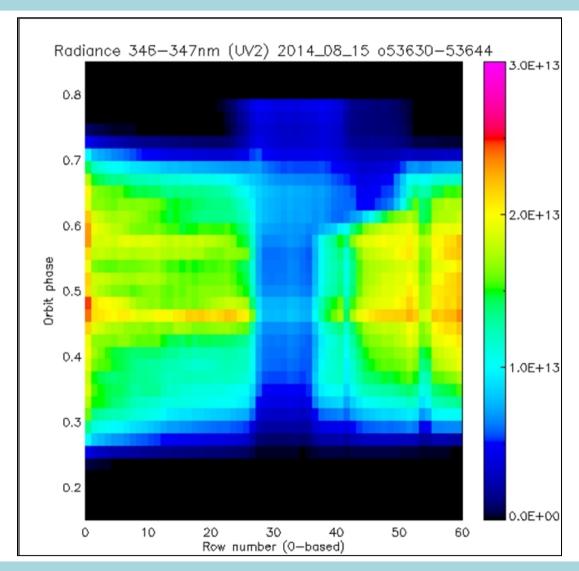
Row anomaly change on August 13, 2014



UV2 radiance daily average



Row anomaly change on August 13, 2014

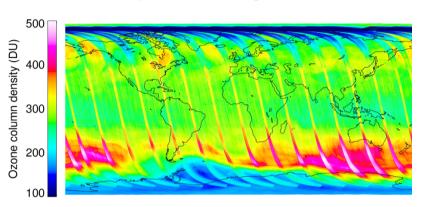


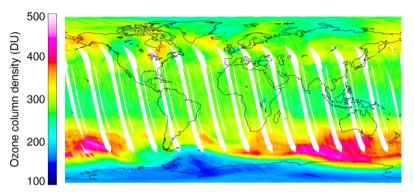
UV2 radiance daily average



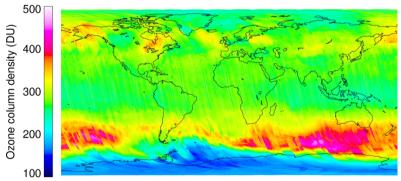
Row anomaly: impact

Now daily coverage within 2 instead of 1 day





OMI ozone column density for September 22nd, 2012. Left: unmasked, right masked. There is no daily global coverage anymore due to the row anomaly.



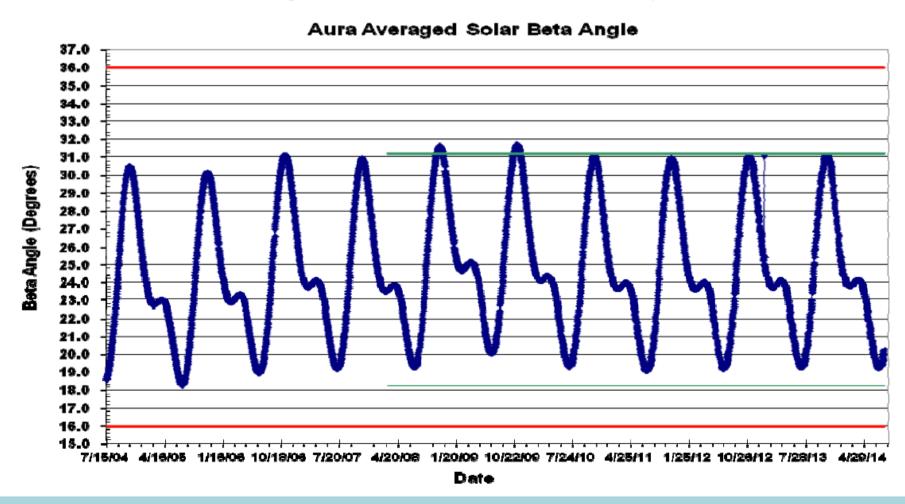
OMI ozone column density for September 22nd plus 23rd, 2012, masked for the row anomaly. This picture shows that global coverage is obtained within 2 days.

Spacecraft Status



Solar beta angle

The solar beta angle is maintained within required limits.

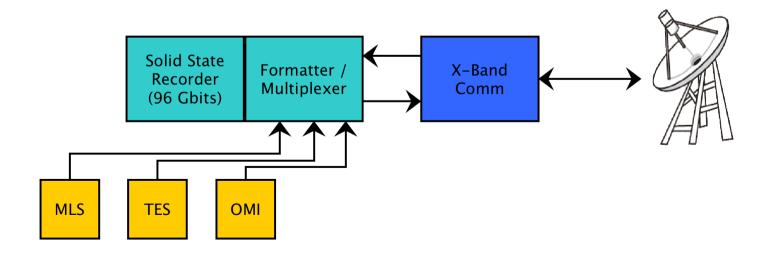


Spacecraft Status



FMU anomaly

- A FMU anomaly started on Dec. 5th, 2007 and is ongoing
- The anomaly causes no science data loss for OMI so far.
- The anomaly impacts the attitude and ephemeris data needed for the L0 -> L1b data processing
- A OMI specific work-around has been implemented: use predicted instead of definitive ancillary data. This works very well.

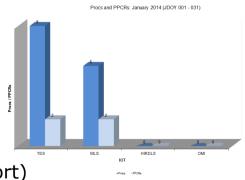


Operations Status



General

- Almost all measurements (>99%) were according to Nominal Operations Baseline.
- WLS voltage yellow/red upperlimits updated after yellow limit violation
- Permanent data loss for orbit 52573 (June 3rd, 2014) due to Poker Flats anomaly.
- Occasional re-scheduling of solar measurements is needed due to S/C thruster maneuvers: avoid contamination of optical parts.
 - MCL for DOY2013/316 was terminated: no time left for rescheduling due to DAM. No loss of science data, only loss of quality.
- Instrument fully operational for more than 99.99% of its time.
- No real-time commanding was needed (last time was on June 27, 2012 for resetting counters after a multi-bit error in IAM memory)

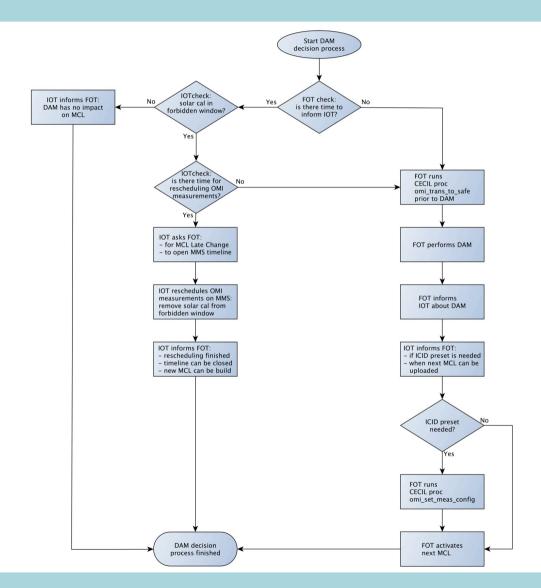


Command activity for January 2014 (from the Aura Monthly Status Report)

Operations Status



New DAM procedure



The OMIS-IOT has proposed a new procedure to be followed in case of a Debris Avoidance Maneuver.

Conclusion



- The instrument status is very good
- The instrument degradation is very slow
- No issues (except for the row anomaly)
- Science data is of very high quality

OUTLOOK

The Instrument Operations Team expects to operate the instrument without any problems for the next coming years.



Celebrations at KNMI



OMI 10 years





BACKUP SLIDES

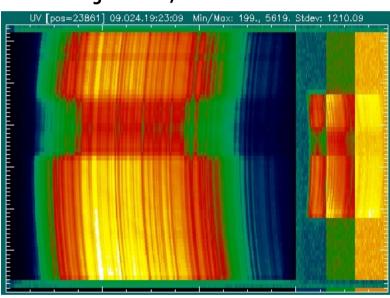


Row anomaly change on Jan 24, 2009

24jan09/orbit 24092

VV [pos=21150] 09.024.17;44:14 Min/Max: 200., 5602. Stdev: 1229.33

24jan09/orbit 24093



OMI LO UV CCD images

wavelength



White Light Source (WLS) voltage alarm

- On November 23, 2011 there was a upper yellow limit violation of the WLS voltage.
- Cause: the WLS filament is becoming thinner over time, its resistance is therefore increasing resulting in a higher voltage over the filament.
- After consulting the instrument builder it was decided to increase the voltage upper yellow and red limits:
 - yellow 13.25 -> 13.75 Volt (will be reached beginning 2020)
 - red 13.50 -> 14.00 Volt

This was implemented on December 8, 2011

